

Research Article

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Two-phase numerical simulation of thermal and solutal transport exploration of a non-Newtonian nanomaterial flow past a stretching surface with chemical reaction

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Abstract: Non-uniform heat sources and sinks are used to control the temperature of the reaction and ensure that it proceeds at the desired rate. It is worldwide in nature and may be found in all engineering applications such as nuclear reactors, electronic devices, chemical reactors, etc. In food processing, heat is used to cook such as microwave ovens, pasteurize infrared heaters, and sterilize food products. Non-uniform heat sources are mainly used in biomedical applications, such as hyperthermia cancer treatment, to target and kill cancer cells. Because of its ubiquitous nature, the idea is taken as our subject of study. Heat and species transfer analysis of a non-Newtonian fluid flow model

under magnetic effects past an extensible moving sheet is modelled and examined. Homogeneous chemical reaction inside the fluid medium is also investigated. This natural phenomenon is framed as a set of Prandtl boundary layer equations under the assumed convective surface boundary constraint. Self-similarity transformation is employed to convert framed boundary layer equations to ordinary differential equations. The resultant system is solved using the efficient finite difference utilized Keller box method with the help of MATLAB programming. The influence of various fluid-affecting parameters on fluid momentum, energy, species diffusion and wall drag, heat, and mass transfer coefficients is studied. Accelerating the Weissenberg number decelerates the fluid velocity. The temperature of the fluid rises due to variations in the non-uniform heat source and sink parameters. Ohmic dissipation affects the temperature profile significantly. Species diffusion reduces when thermophoresis parameter and non-uniform heat source and sink parameters vary. The Eckert number enhances the heat and diffusion transfer rate. Increasing the chemical reaction parameter decreases the shear wall stress and energy transmission rate while improving the diffusion rate. The wall drag coefficient and Sherwood number decrease as the thermophoretic parameter increases whereas the Nusselt number increases. We hope that this work will act as a reference for future scholars who will have to deal with urgent problems related to industrial and technical enclosures.

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1 Introduction

Non-Newtonian fluid models have a wide range of applications, spanning various industries and domains. Non-Newtonian models help create these textures by predicting how ingredients

interact and contribute to food products like sauces, ice cream, and yogurt; fluids, like emulsions and suspensions, need better stability and longer shelf life. Controlled release of medications often relies on gels, creams, or ointments with specific viscoelastic properties. Creating lotions, shampoos, and other products with optimal spreadability, consistency, and sensory experience often involves non-Newtonian fluids. The Williamson fluid model is one amongst this category that finds applications in various fields due to its ability to capture the shear-thinning behaviour of many real-world fluids. The model can be used to understand the behaviour of shear-thinning lubricants under different operating conditions, leading to improved lubricant design and performance. Formulating cosmetics with desired rheological properties often involves using Williamson fluid models to predict their flow and sensory characteristics. The model can be used to simulate the flow of mudslides, slurries, and other non-Newtonian fluids in the environment, aiding in risk assessment and mitigation strategies. Hamid *et al.* [1] investigated the MHD Blasius flow across a vertical plate of radiative Williamson nanofluid. Amanulla and Wakif [2] examined the Williamson fluid numerically under convective heating and radiation effects. Jalili *et al.* [3] explored a thermal study on Williamson fluid over a stretched plate under Lorentz force. Asjad *et al.* [4] investigated the effect of MHD and activation energy on Williamson fluid flow including bioconvection. Malik *et al.* [5] studied the impact of Williamson fluid flow across a 3D linear stretching surface. Maaitah *et al.* [6] analysed the viscous dissipation influence on Williamson fluid over a horizontally saturated porous plate. Sreenivasulu *et al.* [7] explored internal friction impact on non-Newtonian fluid flow. Ramesh *et al.* [8] explored radiative analysis of the nano Carreau-fluid model. Mebarek-Oudina *et al.* [9] investigated the influence of hybrid magneto-convective flow immersed in a porous medium.

The design and analysis of non-uniform heat sources and sinks is a complex topic that requires a good understanding of heat transfer principles. However, the potential benefits of using non-uniform heat sources and sinks are significant, and they are becoming increasingly important in a wide variety of applications. It includes thermal energy storage systems that use non-uniform heat sources and sinks to store and release thermal energy. Concentrated solar power systems use non-uniform heat sources to generate steam for electricity production. In spacecraft, non-uniform heat sources and sinks are used to manage the thermal environment of the spacecraft and its components. This generation concept is to enhance the fluid conductivity while the other reduces fluid energy. Its term is inevitable only when there is a huge temperature difference and has greater importance in MHD flows. These applications were well discussed by the researchers in their early works [10–12]. Konda *et al.* [13] investigated the effects of varied heat sink and source on

non-Newtonian fluid. Jyotshna *et al.* [14] extended the same effect over gallium nitride nanoparticles using the Williamson fluid model. Song *et al.* [15] investigated the same effects on stretched cylinders. Swain *et al.* [16] conducted the same investigation on a porous medium. Sajid *et al.* [17] discussed the effect of an inconstant heat source (sink) on viscous radiative Sutterby nanofluid past the permeable rotative cone. Hussain *et al.* [18] investigated the effects of a heat source (sink) on hybrid nanoflow over a solid stretchable sheet.

From seemingly ordinary activities such as cooking to groundbreaking advancements in medicine and energy, chemical reactions play an essential role in creating our lives and the world around us. Here are some key areas: Fossil fuels like coal, oil, and natural gas are burned in power plants to generate electricity through combustion reactions. To produce energy, nuclear fission and fusion power plants utilize controlled nuclear reactions. Photovoltaic cells use photochemical reactions. These electrochemical devices utilize chemical reactions to convert chemical energy into electrical energy. Poornima *et al.* [19,20] discussed the chemical reaction impacts on MHD flow stretchable surfaces and also on a circular cylinder. Malik *et al.* [21] analysed homogeneous–heterogeneous reactions in the Williamson fluid model across a stretched cylinder. Sarfraz and Masood [22] studied heat transport analysis for nanofluid flows induced by a moving plate with the Cattaneo–Christov double diffusion. Shah *et al.* [23] examined the effect of homogeneous chemical reactions on mixed convective Williamson fluid passing through a penetrable porous wedge. Alrihieli *et al.* [24] discussed MHD dissipative Williamson nanofluid flow with chemical reactions caused by a slippery elastic sheet. Gautam *et al.* [25] considered activation energy and binary chemical reaction impact on MHD flow of Williamson nanofluid in the Darcy–Forchheimer porous medium.

Ohmic dissipation refers to the conversion of electrical energy into thermal energy due to the resistance of a conductor behind heating elements in devices like. The resistance of the wire element converts electrical energy into heat, warming the surrounding environment examples of devices such as toasters, hairdryers, and electric heaters work on the Ohmic dissipation principle. Generating high currents which suffice to melt the fuse is another example. Re-entry vehicles experience significant aerodynamic heating due to friction. This heat can be partially mitigated by using ohmic heating of the vehicle skin to radiate heat away. Ohmic heating is used in some hyperthermia cancer treatment protocols to target and destroy cancer cells. Sreenivasulu *et al.* [26] examined the Ohmic heating impact on non-Newtonian fluid flow. Rashad *et al.* [27] described the Joule heating impact on MHD Williamson hybrid nanoflow. Majid *et al.* [28] investigated the

Ohmic heating impact of a mixed convective flow of Williamson fluid with thermal radiations. Dissipation and heat source/sink impact on Williamson fluid with suction was studied by Hussain *et al.* [29].

With the above research knowledge, no study is focussed on the collective phenomenon of heat and mass transfer analysis of the non-Newtonian Williamson nanofluid model incorporating the asymmetric heat generation/absorption and ohmic dissipation with chemical reactions. Thus, this article focuses on that part of the study considering Williamson nanoflow past a stretching sheet with chemical reaction, non-uniform source of energy, radiation, and Ohmic dissipation, taking into the molecular study of nanoparticles, *i.e.* Brownian motion and thermophoresis. Williamson nanoflows with Ohmic dissipation and radiation can be utilized for efficient energy conversion and harvesting. Specifically in microfluidic fuel cells where the flow can be designed to enhance the mixing and mass transport of reactants within microfluidic fuel cells, leading to higher power output. In nanogenerators, the conversion of heat generated through Ohmic dissipation or radiation into electrical energy using nanomaterials can be explored for miniaturized power generation applications. A visual representation is provided of how several relevant physical factors affect the temperature, concentration, and fluid velocity. Additionally, the Sherwood Number, Nusselt Number, and skin friction coefficients are calculated numerically and portrayed.

2 Mathematical formulation of the flow problem

The scenario where a time-independent, laminar, two-dimensional, incompressible, and electrically conducting Williamson nanofluid past a stretching surface with stretching velocity U_w is investigated. The Cauchy stress tensor s for the current fluid model is defined [30] as

$$s = -pI + \tau_1, \quad (1)$$

where $\tau_1 = \left(\mu_\infty + \frac{\mu_0 - \mu_\infty}{1 - \Gamma \dot{y}} \right) A_1$.

Here, μ_0, μ_∞ are zero and infinite viscosities, Γ represents the time constant, and τ_1 denotes the additional stress tensor. Also, A_1 refers to the Rivlin–Erickson tensor, and $\dot{y} = \sqrt{\frac{\pi}{2}}, \pi = \text{trace}(A_1^2)$. If μ_∞ tends to zero, then we obtain $\tau_1 = [\mu_0 A_1 (1 - \Gamma \dot{y})^{-1}]$.

Furthermore, the sheet is stretched along the x -axis and fluid flow occurs due to stretching and thus occupies the $y \geq 0$ region. Since the fluid is electrically conducting, a transverse magnetic field of uniform strength $B = (0, B_0, 0)$ is

taken perpendicular to the flow following Ohm's law. Initially, the surface and the nanofluid flow are maintained constant. Then, due to convection heat transfer with a heat transfer coefficient, T_w rises the temperature instantaneously and is maintained thereafter. The temperature near the wall is under convection at the boundary T_∞ . In addition, radiation and ohmic dissipation are considered for understanding the energy transfer dynamics. The concentration diffusion near the wall is C_w , and far from the wall, it is denoted by the constant ambient concentration C_∞ . The fluid medium is under homogeneous chemical reactions. Convective wall boundary specification is assumed. As the applied magnetic field is much smaller, the induced magnetic effects are denied. Figure 1 depicts the geometry of these flow configuration phenomena. Due to the aforementioned assumptions and using the boundary layer approximations, the following governing equations representing continuity (2), momentum (3), energy (4), and species diffusion or concentration (5) are modelled below, see the study of Nayak and Mishra [31]:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0, \quad (2)$$

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = v \frac{\partial^2 u}{\partial y^2} + \sqrt{2} v \Gamma \frac{\partial u}{\partial y} \frac{\partial^2 u}{\partial y^2} - \frac{\sigma B_0^2}{\rho} u, \quad (3)$$

$$u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \frac{1}{\rho c_p} \left[k + \frac{16\sigma^* T_\infty^3}{3k^*} \right] \frac{\partial^2 T}{\partial y^2} + \frac{q'''}{\rho c_p} + \tau \left[D_B \frac{\partial C}{\partial y} \frac{\partial T}{\partial y} + \frac{D_T}{T_\infty} \left(\frac{\partial T}{\partial y} \right)^2 \right] \quad (4)$$

$$+ \frac{\mu}{\rho c_p} \left(\frac{\partial u}{\partial y} \right)^2 + \frac{\mu}{\rho c_p} \Gamma \left(\frac{\partial u}{\partial y} \right)^3 + \frac{\sigma B_0^2}{\rho c_p} u^2,$$

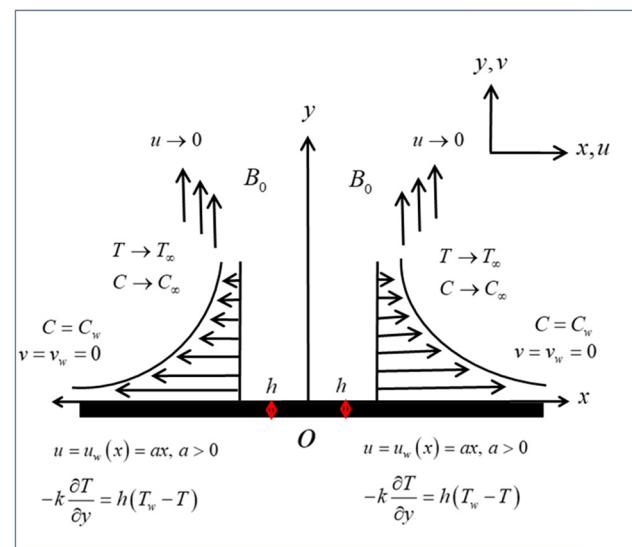


Figure 1: The geometry of the flow model.

$$u \frac{\partial C}{\partial x} + v \frac{\partial C}{\partial y} = D_B \frac{\partial^2 C}{\partial y^2} + \frac{D_T}{T_\infty} \frac{\partial^2 T}{\partial y^2} - K_1(C - C_\infty). \quad (5)$$

In which:

$$q''' = \frac{ku_w(x)}{xv} (Af'(T_w - T_\infty) + B(T - T_\infty)). \quad (6)$$

In addition, the associated physical boundary conditions for the existing model are

$$u = U_w = ax, \quad v = 0, \quad -k \frac{\partial T}{\partial y} = h(T_w - T), \quad C = C_w, \quad (7)$$

when $y = 0$

$u \rightarrow 0, \quad T \rightarrow T_\infty, \quad C \rightarrow C_\infty$ when $y \rightarrow \infty$.

In the above equations, u, v represent velocity components along the x, y directions, v, μ represent the kinematic and dynamic viscosity, σ represents the electrical conductivity, ρ represents the density of the fluid, B_0 represents the uniform magnetic force, K_1 represents the chemical reaction parameter, A, B represent the heat source (sink) parameter, k represents the thermal conductivity, D_B, D_T represent the Brownian motion and thermophoresis coefficients, h represents the heat transfer coefficient, C_p represents the specific heat capacitance, σ^* represents the Stefan Boltzmann constant, k^* represents mean absorption coefficient, and τ signifies the ratio of effective heat capacity and nanofluid heat capacity.

Furthermore, to simplify the scrutiny of the flow, heat, and mass transfer problem, we need to introduce the following similarity transformations [31]:

$$u = u_w f', \quad v = -\sqrt{av} f, \quad \eta = \sqrt{\frac{a}{y}} y, \quad \theta = \frac{T - T_\infty}{T_w - T_\infty}, \quad (8)$$

$$\phi = \frac{C - C_\infty}{C_w - C_\infty}.$$

Here, $f(\eta), \theta(\eta), \phi(\eta)$ are dimensionless stream, temperature, and concentration functions, respectively, where the prime (') denotes the derivative with respect to the pseudo-similarity variable η . Use Eqs. (6) and (8) to convert the governing Eqs. (3)–(5) in the following form:

$$f''' + We f''''' - M(f') + ff'' - (f')^2 = 0, \quad (9)$$

$$\frac{1}{Pr} (1 + Rd) \theta'' + Nb \theta' \phi' + Nt(\theta')^2 + Ec(f'')^2 + EcWe(f'')^3 + MEc(f')^2 + f\theta' + Af' + B\theta = 0, \quad (10)$$

$$\frac{1}{Sc} \phi'' + \frac{1}{Sc} \frac{Nt}{Nb} \theta'' - \gamma \phi + f\phi' = 0. \quad (11)$$

The transformed BCs are

$$f(0) = 0, \quad f'(0) = 1, \quad \theta'(0) = -Bi(1 - \theta(0)), \quad \phi(0) = 1, \quad (12)$$

$$f'(\infty) \rightarrow 0, \quad \theta(\infty) \rightarrow 0, \quad \phi(\infty) \rightarrow 0.$$

In the above equations, the non-dimensional parameters are the magnetic parameter, M , the Weissenberg number, We , the Prandtl number, Pr , the Schmidt number, Sc , the radiation parameter, Rd , the thermophoretic parameter, Nt , the Brownian motion parameter, Nb , the chemical reaction parameter, γ , and the local Reynolds number, Re_x . The definitions of these parameters are mathematically expressed as

$$M = \frac{\sigma B_0^2}{a\rho}, \quad We = \sqrt{2} \Gamma a Re_x^{1/2}, \quad Pr = \frac{v}{a},$$

$$Sc = \frac{v}{D_B}, \quad Rd = \frac{16\sigma^* T_\infty^3}{k^* k}, \quad \gamma = \frac{K_1}{a}, \quad (13)$$

$$Nt = \frac{\tau D_T (T_w - T_\infty)}{v T_\infty}, \quad Nb = \frac{\tau D_B (C_w - C_\infty)}{v},$$

$$Re_x = \frac{u_w x}{v}.$$

The physical quantities are the skin friction coefficient C_{fx} , the local Nusselt number Nu_x , and the local Sherwood number Sh_x , which are defined as follows:

$$C_{fx} = \frac{\tau_w}{\rho u_w^2}, \quad Nu_x = \frac{x q_w}{k(T_w - T_\infty)}, \quad Sh_x = \frac{x M_m}{D_B(C_w - C_\infty)}. \quad (14)$$

At the surface of the sheet, the shear stress, heat flux, and mass flux are mathematically expressed as

$$\tau_w = \mu \left(\frac{\partial u}{\partial y} + \frac{\Gamma}{\sqrt{2}} \left(\frac{\partial u}{\partial y} \right)^2 \right),$$

$$q_w = - \left(k + \frac{16\sigma^* T_\infty^3}{3k^*} \right) \left(\frac{\partial T}{\partial y} \right)_{y=0}, \quad M_m = -D_B \left(\frac{\partial C}{\partial y} \right)_{y=0}. \quad (15)$$

Utilizing Eq. (8) in Eqs. (14) and (15), the resulting reduced form of the gradients are given by

$$Re_x^{1/2} C_{fx} = \left(f''(0) + \frac{We}{2} f'''(0) \right),$$

$$Re_x^{-1/2} Nu_x = -(1 + Rd) \theta'(0),$$

$$Re_x^{-1/2} Sh_x = -\phi'(0). \quad (16)$$

3 Numerical solution procedure of the given scheme

In this section of the work, we need to discuss the complete procedure of the scheme as well as the authentication of the code for the limiting cases with existing work.

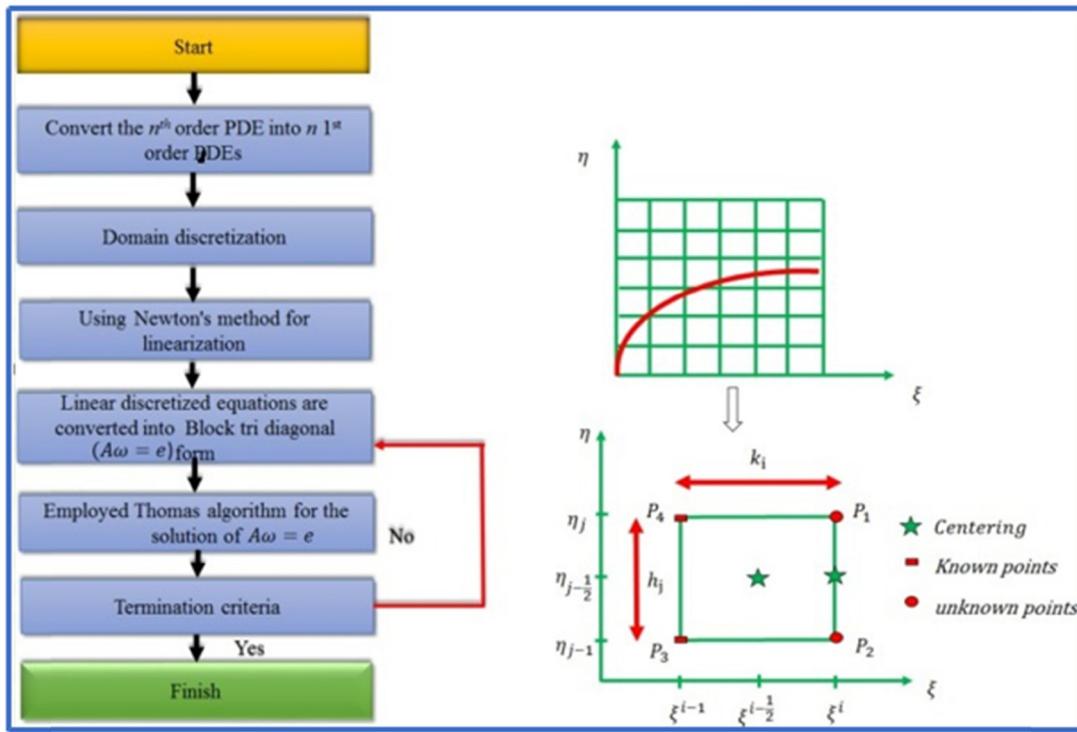


Figure 2: Keller box method.

Following the mathematical modelling, the next step is to build the answer. We opted the Keller–Box approach (KBM), a hidden finite-difference methodology, for the computational solution of the modelled equations since it combines second-degree validity with the ability of step size adaptation. Since its quicker convergence rate relative to conventional numerical techniques, this approach is best suited for solving boundary layer flow problems rather than other explicit techniques such as the RK method, BVP4c, and the shooting technique. Using this approach, higher-order PDEs are reduced to first-order PDEs, which are then translated into central difference formulas. The decomposition of the LU technique is used to solve the matrix–vector form of transformed solutions. The material domain $[0, \infty)$ is used throughout the computation procedure and is condensed to the limited area $[\eta_0, \eta_\infty]$ by altering $\eta_0 = 0$, $\eta_\infty = 20$, $\eta_p = 1,000$, and $h = \frac{\eta_\infty - \eta_0}{\eta_p}$. To establish the initial approximations of the computerized solution, it is needed to increase the total amount of points on the grid by decreasing the step size h . So, that a necessary precision, *i.e.* ε^{-6} will be attained.

Step A: The N th order partial differential equation system is reduced to N first-order equations.

We add the most recent set of variables listed below to convert higher-order ODEs to first-order ODEs such as $P_1(\zeta, \eta)$, $P_2(\zeta, \eta)$, $P_3(\zeta, \eta)$, $P_4(\zeta, \eta)$, $P_5(\zeta, \eta)$, $P_6(\zeta, \eta)$. Let

$$\begin{aligned} f &= f, f' = P_1, P_1' = f'' = P_2, f''' = P_2', \\ \theta &= P_3, P_3' = \theta' = P_4, \theta'' = P_4', \\ \phi &= \phi, \phi' = P_5, \phi'' = P_6. \end{aligned} \quad (17)$$

Implementing Eq. (17) in the similarity equations, we obtain

$$(P_2') + \text{We}(P_2)(P_2') - M(P_1) + (fP_2) - (P_1)^2 = 0, \quad (18)$$

$$\begin{aligned} (P_4') \frac{1}{\text{Pr}} (1 + \text{Rd}) + \text{Nb}(P_4)(P_5) + \text{Nt}(P_4)^2 \\ + \text{Ec}(P_2)^2 + \text{EcWe}(P_2)^3 + \text{MEc}(P_1)^2 + (fP_4) \\ + (AP_1) + B(P_3) = 0, \end{aligned} \quad (19)$$

$$\frac{1}{\text{Sc}}(P_6) + \frac{1}{\text{Sc Nb}}(P_4) - \gamma\phi + f(P_5) = 0. \quad (20)$$

The boundary conditions are

$$\begin{aligned} f(0) &= 0, P_1(0) = 1, P_4(0) = -\text{Bi}(1 - P_3(0)), \phi(0) \\ &= 1 \text{ at } \eta = 0. \end{aligned} \quad (21)$$

$$P_1(\eta) = 0, P_3(\eta) = 0, \phi(\eta) = 0 \text{ as } \eta \rightarrow \infty.$$

Step B: The finite-difference method.

The rectangular net is in the x and y planes, as shown in Figure 2, and the following are the net points:

$$\begin{aligned} \zeta^0 &= 0, \quad \zeta^i = \zeta^{i-1} + k_i, \quad i = 1, 2, 3, \dots, I \\ \eta_0 &= 0, \quad \eta_j = \eta_{j-1} + h_j, \quad j = 1, 2, 3, \dots, J, \end{aligned} \quad (22)$$

where k_i and h_j represent the $\Delta\zeta$ and $\Delta\eta$ spacing, respectively.

$$\left[\frac{\partial(\)}{\partial\zeta} \right]_{j-1/2}^{i-\frac{1}{2}} = \frac{(\)_{j-\frac{1}{2}}^i - (\)_{j-\frac{1}{2}}^{i-1}}{k_i}, \quad (23)$$

$$\left[\frac{\partial(\)}{\partial\eta} \right]_{j-1/2}^{i-\frac{1}{2}} = \frac{(\)_{j-\frac{1}{2}}^i - (\)_{j-\frac{1}{2}}^{i-1}}{h_j}, \quad (24)$$

$$(\)_{j-\frac{1}{2}}^{i-1} = \frac{(\)_{j-1}^{i-1} - (\)_j^i}{2}, \quad (\)_{j-\frac{1}{2}}^i = \frac{(\)_{j-1}^i - (\)_j^i}{2}. \quad (25)$$

The finite-difference form is computed using the central difference technique as follows:

$$f' = P_1 \Rightarrow (P_1)_{j-1/2}^i = \frac{(P_1)_j^i + (P_1)_{j-1}^i}{2} = \frac{(f_j^i - f_{j-1}^i)}{h_j}, \quad (26)$$

$$\begin{aligned} (P_1)' = P_2 \Rightarrow (P_2)_{j-1/2}^i &= \frac{(P_2)_j^i + (P_2)_{j-1}^i}{2} \\ &= \frac{((P_1)_j^i - (P_1)_{j-1}^i)}{h_j}, \end{aligned} \quad (27)$$

$$\begin{aligned} (P_3)' = P_4 \Rightarrow (P_4)_{j-1/2}^i &= \frac{(P_4)_j^i + (P_4)_{j-1}^i}{2} \\ &= \frac{((P_3)_j^i - (P_3)_{j-1}^i)}{h_j}, \end{aligned} \quad (28)$$

$$\begin{aligned} (P_4)' = P_5 \Rightarrow (P_5)_{j-1/2}^i &= \frac{(P_5)_j^i + (P_5)_{j-1}^i}{2} \\ &= \frac{((P_4)_j^i - (P_4)_{j-1}^i)}{h_j}, \end{aligned} \quad (29)$$

$$\begin{aligned} (P_5)' = P_6 \Rightarrow (P_6)_{j-1/2}^i &= \frac{(P_6)_j^i + (P_6)_{j-1}^i}{2} \\ &= \frac{((P_5)_j^i - (P_5)_{j-1}^i)}{h_j}. \end{aligned} \quad (30)$$

Eqs. (26)–(30) are centred at the $\left(\zeta^{i-\frac{1}{2}}, \eta_{j-\frac{1}{2}}\right)$ locations, which are shown below:

$$\begin{aligned} &\left[\frac{(P_2)_j^i - (P_2)_{j-1}^i}{h_j} \right] + \text{We}(P_2) \left[\frac{(P_2)_j^i - (P_2)_{j-1}^i}{h_j} \right] \\ &+ M(P_1)_{j-\frac{1}{2}}^i + (fP_2)_{j-\frac{1}{2}}^i - ((P_1)_{j-1/2}^i)^2 = 0, \end{aligned} \quad (31)$$

$$\begin{aligned} &\left[\frac{(P_4)_j^i - (P_4)_{j-1}^i}{h_j} \right] \frac{1}{\text{Pr}} (1 + \text{Rd}) + \text{Nb} (P_4 P_5)_{j-\frac{1}{2}}^i + \text{Nt} (P_4^2)_{j-\frac{1}{2}}^i \\ &+ \text{Ec} (P_2^2)_{j-\frac{1}{2}}^i + \text{EcWe} (P_2^3)_{j-\frac{1}{2}}^i + M \text{Ec} ((P_1)_{j-1/2}^i)^2 \end{aligned} \quad (32)$$

$$\begin{aligned} &+ M \text{Ec} ((E_1)^2)_{j-1/2}^i + (fP_4)_{j-\frac{1}{2}}^i + A (P_1)_{j-\frac{1}{2}}^i \\ &+ B (P_3)_{j-1/2}^i = 0, \end{aligned}$$

$$\begin{aligned} &\frac{1}{\text{Sc}} \left[\frac{(P_6)_j^i - (P_6)_{j-1}^i}{h_j} \right] + \frac{1}{\text{Sc Nb}} \left[\frac{(P_4)_j^i - (P_4)_{j-1}^i}{h_j} \right] \\ &- \gamma (\phi)_{j-\frac{1}{2}}^i + f (P_5)_{j-\frac{1}{2}}^i = 0, \end{aligned} \quad (33)$$

with boundary conditions:

$$\begin{aligned} f_0^i &= 0, \quad (P_1)_0^i = 1, \quad \phi_0^i = 0, \quad (P_4)_0^i = -\text{Bi}(1 - \theta_0) \\ (P_1)_j^i &= 0, \quad (\theta)_j^i = 0, \quad (\phi)_j^i = 0. \end{aligned} \quad (34)$$

Step C: Newton's linearization approach.

Using well-known techniques $(f_j^{n-1}, (P_1)_j^{n-1}, (P_2)_j^{n-1}, (P_3)_j^{n-1}, (P_4)_j^{n-1}, (P_5)_j^{n-1} \dots)$, the unknown $(f_j^n, (P_1)_j^n, (P_2)_j^n, (P_3)_j^n, (P_4)_j^n, (P_5)_j^n \dots)$ are predicted to be $0 \leq j \leq J$.

$$\begin{aligned} &(f_j^n, (P_1)_j^n, (P_2)_j^n, (P_3)_j^n, (P_4)_j^n \dots) \\ &\equiv (f_j, (P_1)_j, (P_2)_j, (P_3)_j, (P_4)_j \dots). \end{aligned} \quad (35)$$

The collection of equations of central difference is denoted as

$$\frac{(P_1)_j + (P_1)_{j-1}}{2} = \frac{f_j - f_{j-1}}{h_j}, \quad (36)$$

$$\frac{(P_2)_j + (P_2)_{j-1}}{2} = \frac{(P_1)_j + (P_1)_{j-1}}{h_j}, \quad (37)$$

$$\frac{(P_4)_j + (P_4)_{j-1}}{2} = \frac{(P_3)_j + (P_3)_{j-1}}{h_j}, \quad (38)$$

$$\frac{(P_6)_j + (P_6)_{j-1}}{2} = \frac{(P_5)_j + (P_5)_{j-1}}{h_j}. \quad (39)$$

We present the iterates below to use Newton's method to convert to a nonlinear collection of equations approach:

$$\Delta f_j^{(n)} + f_j^n = f_j^{(n+1)}, \quad (40)$$

$$\Delta (P_1)_j^{(n)} + (P_1)_j^{(n)} = (P_1)_j^{(n+1)}, \quad (41)$$

$$\Delta(P_2)_j^{(n)} + (P_2)_j^{(n)} = (P_2)_j^{(n+1)}, \quad (42)$$

$$\Delta(P_3)_j^{(n)} + (P_3)_j^{(n)} = (P_3)_j^{(n+1)}, \quad (43)$$

$$\Delta(P_4)_j^{(n)} + (P_4)_j^{(n)} = (P_4)_j^{(n+1)}, \quad (44)$$

$$\Delta(P_5)_j^{(n)} + (P_5)_j^{(n)} = (P_5)_j^{(n+1)}. \quad (45)$$

This method results in the following linear system (the superscript $[n]$ has been deleted for clarity):

$$\Delta f_j - \Delta f_{j-1} - \frac{h_j}{2} \Delta(P_1)_j - \frac{h_j}{2} \Delta(P_1)_{j-1} - (l_1)_j = 0, \quad (46)$$

$$\Delta(p_1)_j - \Delta(P_1)_{j-1} - \frac{h_j}{2} \Delta(P_2)_j - \frac{h_j}{2} \Delta(P_2)_{j-1} - (l_2)_j = 0, \quad (47)$$

$$\Delta(P_3)_j - \Delta(P_3)_{j-1} - \frac{h_j}{2} \Delta(P_4)_j - \frac{h_j}{2} \Delta(P_4)_{j-1} - (l_3)_j = 0, \quad (48)$$

$$\Delta(P_5)_j - \Delta(P_5)_{j-1} - \frac{h_j}{2} \Delta(P_6)_j - \frac{h_j}{2} \Delta(P_6)_{j-1} - (l_4)_j = 0, \quad (49)$$

$$(G_1)_j \Delta(P_2)_j + (G_2)_j \Delta(P_2)_{j-1} + (G_3)_j \Delta(f_j) + (G_4)_j \Delta f_{j-1} + (G_5)_j \Delta(P_1)_j + (G_6)_j \Delta(P_1)_{j-1} = 0, \quad (50)$$

$$\begin{aligned} & (H_1)_j \Delta(p_4)_j + (H_2)_j \Delta(p_4)_{j-1} + (H_3)_j \Delta(f_j) + (H_4)_j \Delta(f_{j-1}) \\ & + (H_5)_j \Delta(P_1)_j + (H_6)_j \Delta(P_1)_{j-1} + (H_7)_j \Delta(p_5)_j \\ & + (H_8)_j \Delta(p_5)_{j-1} + (H_9)_j \Delta(P_2)_j + (H_{10})_j \Delta(P_2)_{j-1} \\ & + (H_{11})_j \Delta(P_3)_j + (H_{12})_j \Delta(P_3)_{j-1} + (-l_6)_j = 0, \end{aligned} \quad (51)$$

$$\begin{aligned} & (I_1)_j \Delta(P_4)_j + (I_2)_j \Delta(P_4)_{j-1} + (I_3)_j \Delta(P_5)_j + (I_4)_j \Delta(P_5)_{j-1} \\ & + (I_5)_j \Delta(P_6)_j + (I_6)_j \Delta(P_6)_{j-1} + (I_7)_j \Delta(\phi)_j \\ & + (I_8)_j \Delta(\phi)_{j-1} - (l_7)_j = 0, \end{aligned} \quad (52)$$

with the boundary conditions :

$$\begin{aligned} \Delta f_0 &= 0, \Delta(P_1)_0 = 1, \Delta(\phi)_0 = 1, \Delta(P_4)_0 = \text{Bi} \Delta \theta_0 \text{ at } \eta = 0, \\ \Delta(P_1)_J &= 0, \Delta(\phi)_J = 0, \Delta(\theta)_J = 0 \text{ as } \eta \rightarrow \infty. \end{aligned} \quad (53)$$

Step D: Tridiagonal system of solution.

The block-elimination approach may be used to solve the linearized difference equations (46)–(53), according to Cebeci and Bradshaw [32], because the system is block-tridiagonal in structure. The block-tridiagonal structure is often made of variables or constants, but in this case, an unusual aspect is that it is made of block matrices, Eqs. (46)–(53) may be represented in matrix–vector form as

$$Q\Omega = l, \quad (54)$$

where

$$Q = \begin{bmatrix} [Q_1] & [R_1] \\ [R_2] & [Q_2] & [S_2] \\ & \ddots & \ddots \\ & & [R_{j-1}] & [Q_{j-1}] & [S_{j-1}] \\ & & & [R_j] & [Q_j] \end{bmatrix},$$

$$\Omega = \begin{bmatrix} [\Omega_1] \\ [\Omega_2] \\ \vdots \\ [\Omega_{j-1}] \\ [\Omega_j] \end{bmatrix} \text{ and } l = \begin{bmatrix} [l_1] \\ [l_2] \\ \vdots \\ [l_{j-1}] \\ [l_j] \end{bmatrix}.$$

The matrix components are the following:

$$[Q_1] = \begin{bmatrix} 0 & 0 & 0 & 1 & 0 & 0 & 0 \\ -z & 0 & 0 & 0 & -z & 0 & 0 \\ 0 & -z & 0 & 0 & 0 & -z & 0 \\ 0 & 0 & -z & 0 & 0 & 0 & -z \\ G_2 & 0 & 0 & G_3 & G_1 & 0 & 0 \\ H_6 & H_2 & H_4 & H_9 & H_5 & H_1 & H_3 \\ 0 & C_4 & C_2 & C_7 & 0 & C_3 & C_1 \end{bmatrix},$$

$$[Q_j] = \begin{bmatrix} -z & 0 & 0 & 1 & 0 & 0 & 0 \\ -1 & 0 & 0 & 0 & -z & 0 & 0 \\ 0 & -1 & 0 & 0 & 0 & -z & 0 \\ 0 & 0 & -1 & 0 & 0 & 0 & -z \\ G_6 & 0 & 0 & G_3 & G_1 & 0 & 0 \\ H_8 & H_{12} & 0 & H_9 & H_5 & H_1 & H_3 \\ 0 & 0 & C_6 & C_7 & 0 & C_3 & C_1 \end{bmatrix},$$

$$[R_j] = \begin{bmatrix} 0 & 0 & 0 & -1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & -z & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & -z & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & -z \\ 0 & 0 & 0 & G_4 & G_2 & 0 & 0 \\ 0 & 0 & 0 & H_{10} & H_6 & H_2 & 0 \\ 0 & 0 & 0 & C_8 & 0 & C_4 & C_2 \end{bmatrix},$$

$$[S_j] = \begin{bmatrix} -z & 0 & 0 & 0 & 0 & 0 & 0 \\ 1 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & -z & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 & 0 \\ G_5 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & H_3 & H_{11} & 0 & 0 & 0 & 0 \\ C_7 & C_9 & C_{11} & 0 & 0 & 0 & 0 \end{bmatrix}.$$

For the category $J \geq j \geq 2$:

$$[\Delta_1] = [\Delta(P_4)_0 \Delta(P_3)_0 \Delta(P_2)_0 \Delta(P_5)_0 \Delta f_1 \Delta(P_4)_1 \Delta(P_4)_1 \Delta(P_4)_1]^T,$$

for $2 \leq j \leq J$,

$$[\Delta_j] = [\Delta(P_4)_{j-1} \Delta(P_3)_{j-1} \Delta(P_2)_{j-1} \Delta(P_5)_j \Delta f_j \Delta(P_4)_j \Delta(P_4)_j \Delta(P_4)_j]^T,$$

for $1 \leq j \leq J$,

$$[l_j] = \begin{bmatrix} (l_1)_{j-\frac{1}{2}} \\ (l_2)_{j-\frac{1}{2}} \\ (l_3)_{j-\frac{1}{2}} \\ (l_4)_{j-\frac{1}{2}} \\ (l_5)_{j-\frac{1}{2}} \\ (l_6)_{j-\frac{1}{2}} \\ (l_7)_{j-\frac{1}{2}} \\ (l_8)_{j-\frac{1}{2}} \end{bmatrix}.$$

The *LU* decomposition technique may be used to find the solution of a tri-diagonal system. Assuming that matrix A is non-singular, it may be factored into the product of two matrices, denoted by the notation $A = LU$

$$L = \begin{bmatrix} [a_1] & & & & & & & \\ [b_1] & [a_2] & & & & & & \\ & [b_3] & [a_3] & \cdots & & & & \\ & & & & [b_{J-1}] & [a_{J-1}] & & \\ & & & & & [b_J] & [a_J] & \\ & & & & & & & \end{bmatrix},$$

$$U = \begin{bmatrix} [I_1] & [c_1] & & & & & & \\ [I_2] & [c_2] & & & & & & \\ & \cdots & & & & & & \\ & & [I_{J-1}] & [c_{J-1}] & & & & \\ & & & [I_J] & & & & \end{bmatrix},$$

where $[a_j]$ and $[c_j]$ are (8×8) matrices, and $[I_j]$ is the (8×8) identity matrix. We may obtain the elements of these matrices using the following equations:

$$[A_1] = [a_1], \quad (55)$$

$$[A_j] = [b_j][c_j] + [a_j] \quad \text{for } j = 2, 3, \dots, J, \quad (56)$$

$$[C_j] = [a_j][c_j], \quad \text{for } j = 1, 2, 3, \dots, J-1, \quad (57)$$

$$[B_j] = [b_j], \quad \text{for } j = 2, 3, 4, \dots, J. \quad (58)$$

Hence, using *LU* decomposition, the above system is solved.

4 Results and discussion

The obtained numerical results are then plotted for understanding the physical phenomena of various flow affecting parameters on the physical and engineering quantities

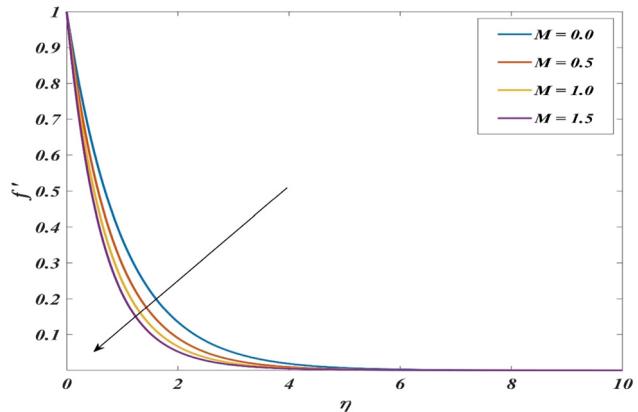


Figure 3: Variation of f' for various values of M .

presented in the form of graphs (Figures 3–20). Here, the effects of dimensionless parameters on the fluid velocity (f'), temperature (θ), and species concentration (ϕ) are very crucial. The parameters are the magnetic parameter, M , the Weissenberg number, We , the Prandtl number, Pr , the Schmidt number, Sc , the radiation parameter, Rd , the thermophoretic parameter, Nt , the Brownian motion parameter, Nb , the chemical reaction parameter, γ , and the irregular heat source/sink parameters, A, B . The findings of the current investigation were achieved for fixed values of $Pr = 0.7$, $M = 0.5$, $We = 0.1$, $Sc = 1.0$, $Ec = 0.1$, $A = B = 0.1$, $Nt = Nb = 0.1$, $n_p = 2, 500$, and $\eta_{\max} = 20$.

Table 1 shows the values of shear stress, heat, and mass transfer rates for different grid points. It is seen that for grid points above 20, a convergency in the numerical values is noted for all engineering quantities such as wall friction coefficient, energy, and mass gradient. Table 2 is the comparison of our numerical computation with the

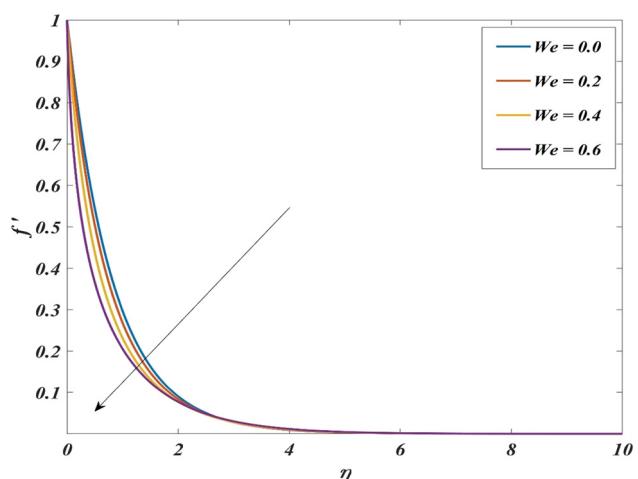
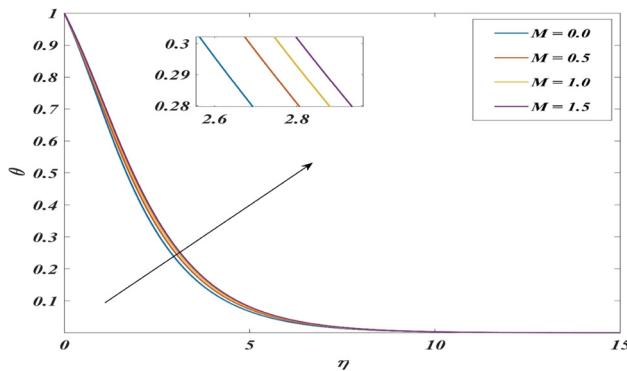
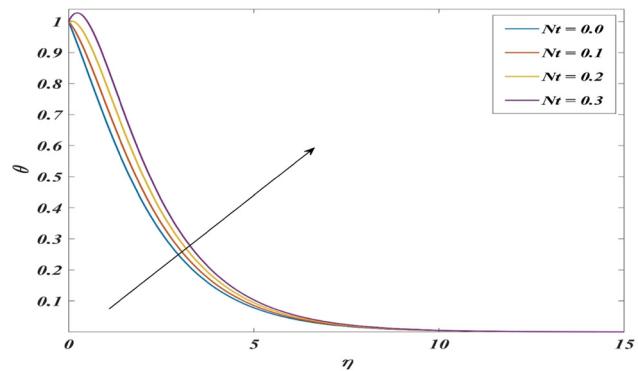
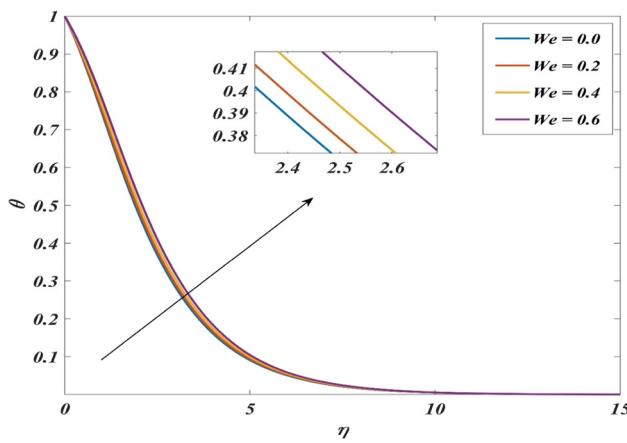
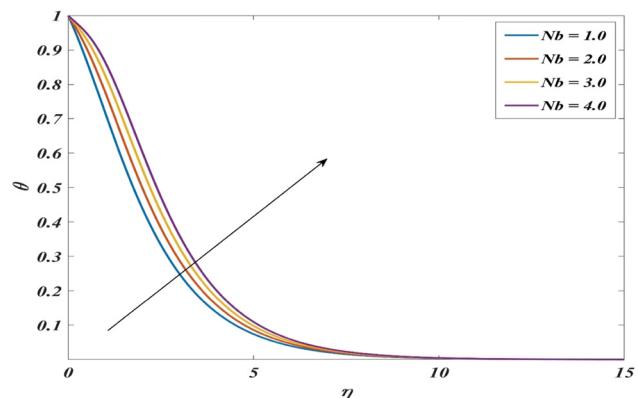
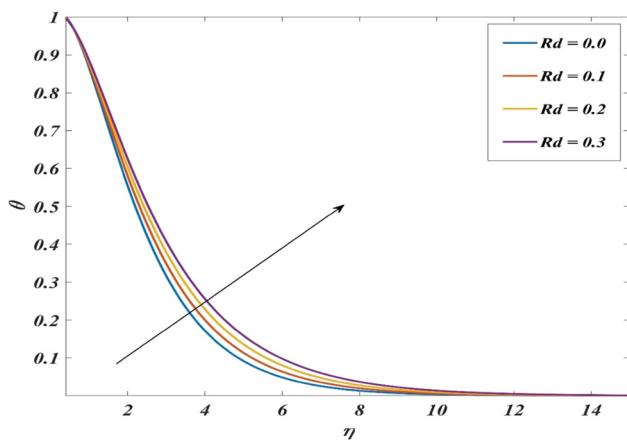


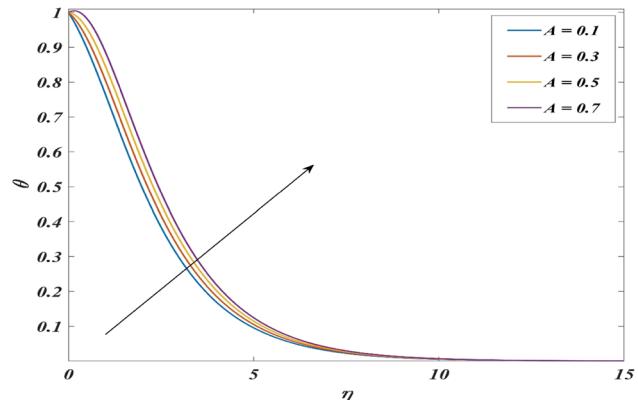
Figure 4: Variation of f' for various values of We .

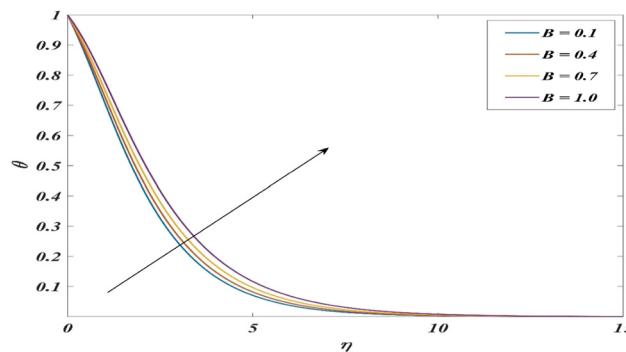
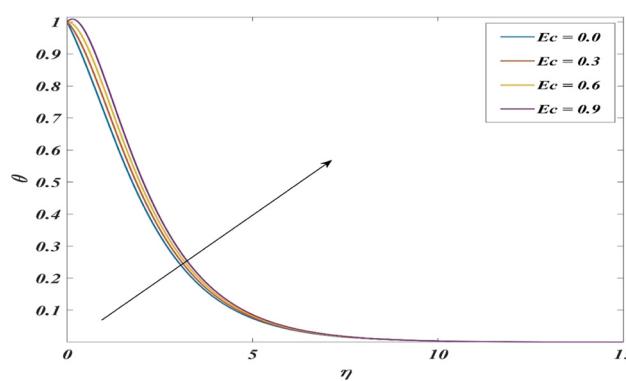
Figure 5: Variation of θ for various values of M .Figure 8: Variation of θ for various values of Nt .Figure 6: Variation of θ for various values of We .Figure 9: Variation of θ for various values of Nb .Figure 7: Variation of θ for various values of Rd .

existing results [33,34]. An excellent agreement is found between our work and them for the limiting cases.

Figure 3 shows the influence of M ($M = 0.0, 0.5, 1.0, 1.5$) on motion of the velocity profile. A magnetic field exerts a force, known as the Lorentz force, on a moving

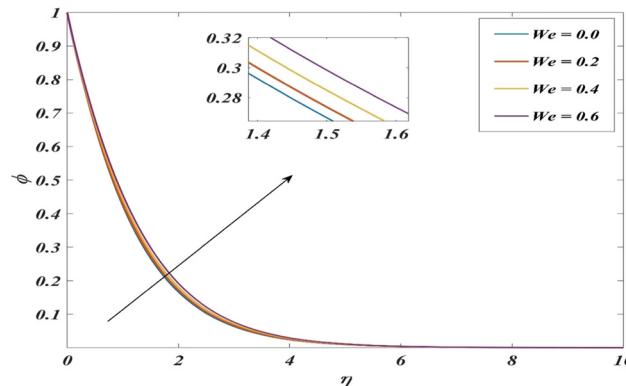
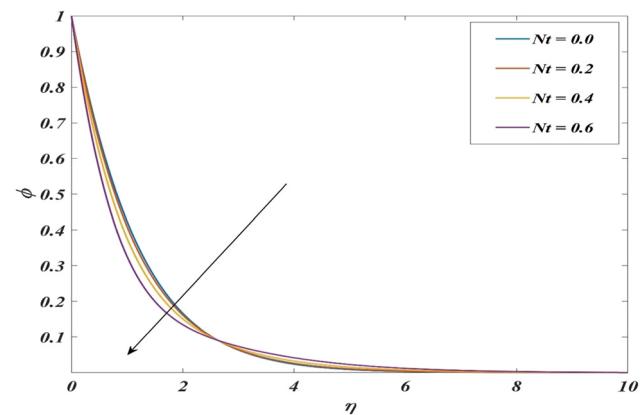
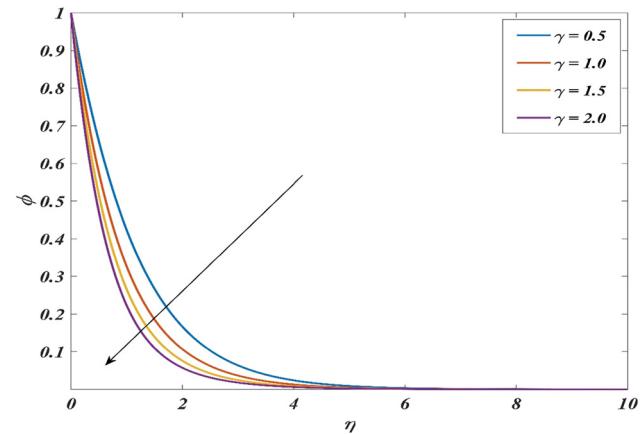
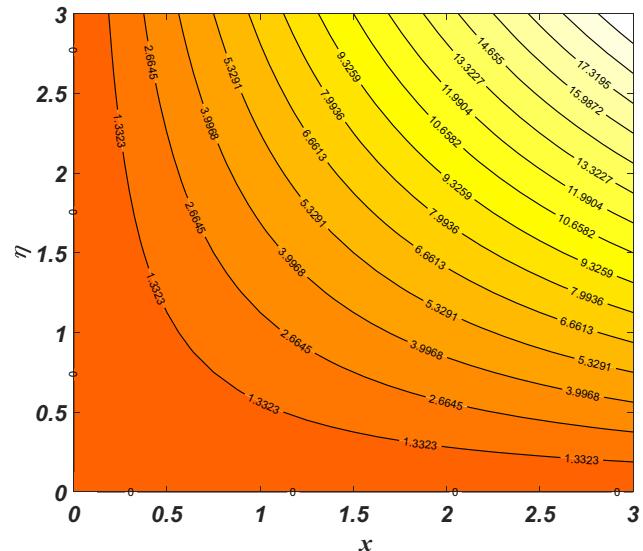
charged particle. This force acts perpendicular to both the particle's velocity and the direction of the magnetic field itself. This Lorentz force resists the fluid motion. Figure 4 depicts the impact of the Weissenberg number ($We = 0.0, 0.1, 0.3, 0.5$) on the field of flow speed. An increase in the Weissenberg number leads to a decrease in the velocity of the fluid. This is because the elastic nature of the fluid or

Figure 10: Variation of θ for various values of A .

Figure 11: Variation of θ for various values of B .Figure 12: Variation of θ for various values of Ec .

the delayed response to shear forces creates resistance to flow.

Figures 5–13 show the temperature profiles for different values of Weissenberg number, Eckert number, and Prandtl number, thermophoresis and Brownian motion parameter, Schmidt number, reaction rate parameter, radiation parameter, and non-uniform heat source (sink) parameters, respectively. For simulation purposes, the standard fixed values for these parameters are ($Pr = 0.7$, $M = 0.5$, $Sc = 1.0$, $\gamma = 0.5$, $A = B = 0.1$, $Nt = Nb = 0.1$, and $Rd = 0.5$).

Figure 13: Variation of ϕ for various values of We .Figure 14: Variation of ϕ for various values of Nt .Figure 15: Variation of ϕ for various values of γ .Figure 16: Streamline for $We = 0.1$.

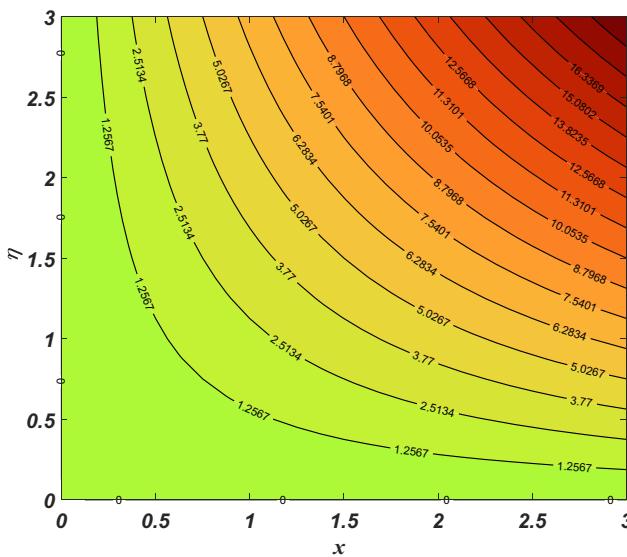


Figure 17: Streamline for $We = 0.3$.

The magnetic field strength application on the electrically conducting fluid makes it more viscous which heats up the fluid. Thus, the temperature profile rises with increasing magnetic numbers (Figure 5). The Lorentz force can also hinder the bulk flow of the nanofluid, particularly at higher magnetic field strengths. This opposition to flow can impede the movement of hot and cold regions within the fluid, potentially reducing the overall heat transfer rate.

Figure 6 describes the temperature profiles for different values for the Weissenberg number. Improving Weissenberg number leads to a rise in the temperature of nanofluids due to the formation of a thicker thermal boundary layer. A thicker boundary layer means the hot

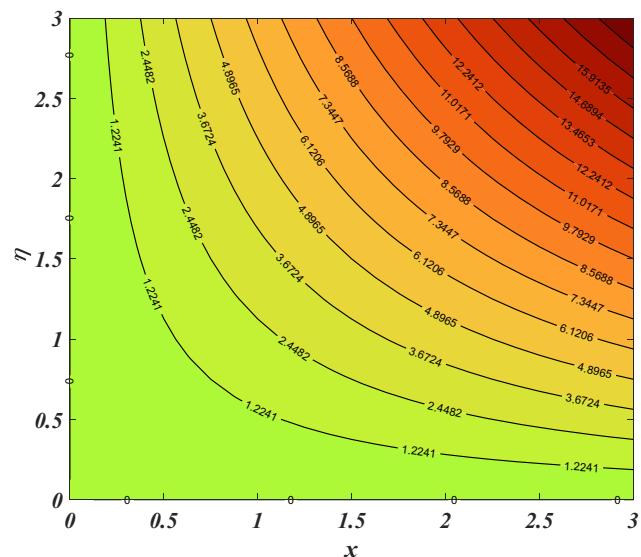


Figure 19: Streamline for $M = 1.0$.

fluid layer close to the wall is confined to a smaller region. Figure 7 shows the temperature profiles for different values for radiation parameters. Increasing the radiation parameter ($Rd = 0.0, 0.5, 1.0, 1.5$) enhances the fluid temperature. The rate of radiative heat transfer increases dramatically with rising temperature. This relationship is governed by the Stefan–Boltzmann Law, which applies to all temperatures above absolute zero. As the temperature of the nanofluid increases, the effect of radiation becomes more dominant compared to conduction.

Figures 8 and 9 show the temperature profiles for different values of the thermophoresis and Brownian motion parameters. Increasing the Nt ($= 0.0, 0.1, 0.2, 0.3$) and

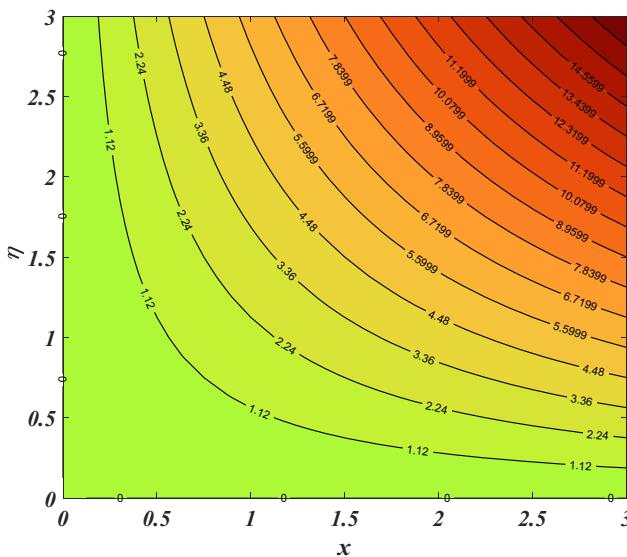


Figure 18: Streamline for $We = 0.5$.

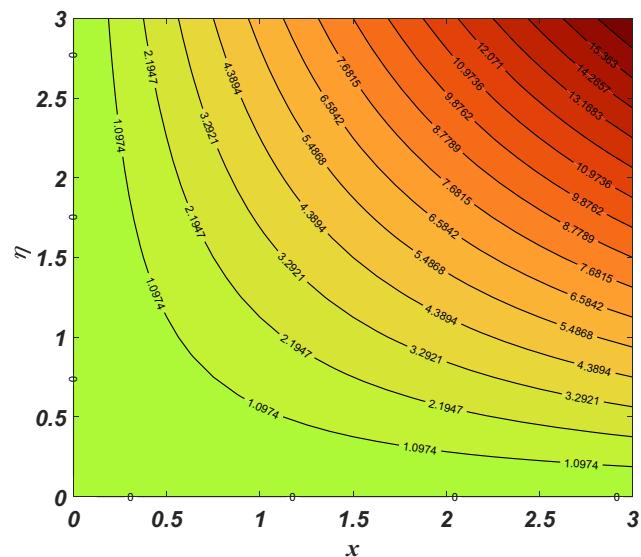


Figure 20: Streamline for $M = 2.0$.

Table 1: Mesh sensitivity test for $-f''(0)$, $-\theta'(0)$, $-\phi'(0)$ with varying values $Pr = 0.7$, $M = 0.5$, $We = 0.1$, $Sc = 1.0$, $Ec = 0.1$, $A = B = 0.1$, $Nt = Nb = 0.1$, $n_p = 2,500$, $\eta_{\max} = 20$

Grid points	KBM	bvp4c	KBM	bvp4c	KBM	bvp4c
	$-f''(0)$	$-f''(0)$	$-\theta'(0)$	$-\theta'(0)$	$-\phi'(0)$	$-\phi'(0)$
20	1.0624	1.1026	0.4675	0.3927	0.6847	0.7148
40	1.0875	1.1026	0.4582	0.3927	0.6957	0.7148
80	1.0954	1.1026	0.4025	0.3927	0.7200	0.7148
160	1.1000	1.1026	0.3990	0.3927	0.7286	0.7148
320	1.1015	1.1026	0.3950	0.3927	0.7325	0.7148
640	1.1020	1.1026	0.3920	0.3927	0.7420	0.7148

Nb ($= 1.0, 2.0, 3.0, 4.0$) values results in enhancement of the fluid temperature. The zig-zag movement of nanoparticles produces microscopic turbulence, which leads to a temperature rise. The same phenomenon is observed in the nanofluid temperature when thermophoresis transports heat from a hotter to a colder region.

Figures 10 and 11 portray the temperature profiles for different values of a non-uniform heat source and sink parameters (A and B). Improving $A, B = (0.0, 0.3, 0.6, 1.0)$ values,

Table 2: Comparing the current findings with those of earlier studies for $-\theta'(0)$ taking $We = M = A = B = Nt = Nb = Ec = 0$

Pr	Khan and Pop [33]	Srinivasulu and Goud [34]	Present
0.2	0.1723	0.1909	0.1900
0.7	0.4539	0.4543	0.4456
2.00	0.9113	0.9121	0.9025
7.00	1.8954	1.9010	1.8952
20.0	3.3539	3.3827	3.2852
70.0	6.4621	6.6473	6.4624

the temperature profiles also increase. The physical interpretation behind this is that the energy source directly heats the fluid, and hence, the local temperature of the nanofluid near the source will rise, and the presence of radiant energy is also responsible for heat enhancement. The nanofluid itself might generate heat due to internal processes such as chemical reactions. This internal heat contributes to the overall temperature rise. Figure 12 shows the temperature profiles for different values for Eckert number (Ec). As the Eckert number ($Ec = 0.1, 0.5, 1.0, 1.5$) improves, the internal friction of the fluid also rises, thereby enhancing the temperature of the fluid.

Table 3: List of numerical values for skin friction coefficient, the local Nusselt number and local Sherwood number such as $Pr = 0.7$, $M = 0.5$, $We = 0.1$, $Sc = 1.0$, $Ec = 0.1$, $A = B = 0.1$, $Nt = Nb = 0.1$, $n_p = 2,500$, $\eta_{\max} = 20$.

We	M	A	B	Ec	γ	Sc	Nt	Nb	Cf _x	Nu _x	Sh _x
0.0									2.14655	0.48412	1.35455
0.4									2.81216	0.48652	1.34594
0.8									2.86745	0.48720	1.34334
	0.0								1.40857	0.32242	1.36996
	0.5								1.71916	0.32356	1.36267
	1.0								1.98263	0.33565	1.35735
		0.0							1.33506	0.67823	1.41617
		0.3							1.33506	0.67956	1.47843
		0.6							1.33506	0.67965	1.54069
			0.0						1.33508	0.68965	1.47898
			0.3						1.33508	0.68985	1.46454
			0.6						1.33508	0.68960	1.45213
				0.1					1.40857	0.76852	1.40290
				0.5					1.40857	0.76952	1.47612
				1.0					1.40857	0.76982	1.58595
					0.1				1.33506	0.45685	1.38037
					0.3				1.17172	0.45582	1.44012
					0.5				1.00838	0.45466	1.55038
						1.0			1.98263	0.56324	1.35735
						1.5			1.98263	0.56325	1.48084
						2.0			1.98263	0.56325	1.58476
							0.2		1.65864	0.62531	1.46521
							0.4		1.65842	0.62542	1.46462
							0.6		1.65625	0.62452	1.45652
								0.2	1.58621	0.56112	1.35621
								0.4	1.58425	0.56021	1.35610
								0.6	1.56822	0.54125	1.34560

Figure 13 shows the Weissenberg influence on species diffusion. Augmenting We ($= 0.0, 0.4, 0.8, 1.2$) increases the species diffusion profiles. Figure 14 presents the parameter Nt ($0.0, 0.2, 0.4, 0.6$) impact on nanoparticle diffusion. It is observed that the diffusion of nanoparticles is decreased as the thermophoresis parameter escalates. As the chemical reaction parameter progresses, the species concentration of reactants decreases. The random movement of nanoparticles can counteract the concentration decrease by causing nanoparticles to move within the fluid, but their effects might be diminished by strong chemical reactions (Figure 15).

Table 3 shows the different engineering quantities affected by the flow parameters. As the Weissenberg number increases, the skin friction coefficient and the Nusselt number increase. But Sherwood's number decreases. The presence of magnetic force reduces the wall friction coefficient and heat transfer coefficient but decreases the concentration gradients. The energy source increases the energy transfer rate while the species diffusion rate decreases. The Eckert number enhances the heat and diffusion transfer rate. Increasing the chemical reaction parameter decreases the shear wall stress and energy transmission rate while the diffusion rate improves. When a species has a lower diffusivity (higher Sc), it is more susceptible to being carried by the flow (convection) rather than relying solely on its random movement (diffusion) to reach its destination. This translates to a more efficient mass transfer process, reflected by a higher Sherwood number. The random motion of nanoparticles decreases all the three engineering quantities. The wall drag coefficient and Sherwood number reduce for increasing thermophoretic parameters, whereas the Nusselt number increases.

Figures 16–18 show the streamlines for $We = (0.1, 0.3, \text{ and } 0.5)$. The starting points ($We = 0.1$) 1.3323 to 18.6518 and ($We = 0.3$) 1.2567 and 17.5963 and ($We = 0.5$) 1.12 and 15.6799. In each contour, it is observed that the amount of flow of energy is uniform, and the contour plot portrays a symmetry at the centre. Figures 19 and 20 show the streamlines for M ($M = 1.0, \text{ and } 2.0$). The starting points are ($M = 1.0$) 1.2241–17.1376 and ($M = 2.0$) 1.0974 and 15.363. The nanoparticles' concentration is high at the centre and the flow of energy is uniform.

5 Conclusion

The purpose of this work was to examine the effects of radiation, viscous dissipation, non-uniform heat source and sink, chemical reaction, and heat and mass transfer influences on non-Newtonian nanofluid flow. To solve the formulated boundary layer equations, the Keller box method

is utilized. The implications of various pertinent parameters on the flow field are numerically computed using MATLAB and portrayed as graphs and tables. These key findings can be summarized as follows:

- Augmenting Weissenberg number and magnetic parameter decelerates the nanofluid velocity.
- Incrementing non-uniform heat source parameters accelerates both fluid velocity and temperature. The temperature intensifies as Eckert's number increases.
- Declining species concentration profiles is noticed when thermophoresis parameter and non-uniform heat source parameters boost.
- As Weissenberg's number increases, the skin friction coefficient and Nusselt number increase, but Sherwood's number decreases.
- Enhancing energy source parameters increases the energy transfer rate, while species diffusion rate decreases. Eckert number enhances the energy and diffusion transfer rate.
- Increasing chemical reaction parameters decreases the shear wall stress and energy transmission rate while the diffusion rate improves.

This problem can also be extended to add different physical impacts such as nanofluids, hybrid nanofluids, ternary hybrid nanofluids, non-Newtonian models and concentration equations, etc. In addition, we can also apply distinct schemes such as ANN, fractional derivatives, and ARA-Sumudu decomposition method [35–39].

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Data availability statement: The datasets used and/or analysed during the current study are available from the corresponding author upon reasonable request.

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